Vibration Reduction in Lightweight Floor/Ceiling Systems with a Sand-Sawdust Damping Layer

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Summary

This paper shows how to use a mathematical model to predict the vibration of lightweight timber-framed floor/ceiling systems (LTFSs) caused by mechanical excitation. The LTFS considered here is made up of an upper layer (including the floor), cavity space with timber joists and the ceiling. These components are joined by timber battens, ceiling furring channels and ceiling clips, which are also included in the model. The vibration in the structure is caused by a localized excitation on the top surface and the resulting vibration level of the ceiling surface will be analysed. The cavity space is filled with fibre infill for damping the sound transmitting through the cavity. A unique feature of the design and the model is the sand-sawdust mixture in the upper layer. The theoretical model and the experimental measurements show that the sand-sawdust dampens the vibration in the frequency range between 10 and 200 Hz. The model uses the classical theories of elastic plates, beams and room acoustics together with the Fourier expansion method for solving the system of partial differential equations. The damping by the sand-sawdust and the fibre infill are found by comparing the numerical simulations against the experimental measurements. This paper will show that the simple linear frequency dependent loss factors can be used in the model to predict the low-frequency vibrations of LTFSs.

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1. Introduction

Most residential buildings can be classified as either concrete-based or lightweight timber-framed systems. The concrete-based systems primarily use concrete slabs for walls and floors, and the lightweight systems use timber-framed composite structures. In this paper we study the sound insulation performances of lightweight timberframed floor/ceiling systems (LTFSs) depicted in Figure 1. Such LTFSs have become popular due to their ease of construction and environmentally friendly use of timber. As the popularity of the systems grows, their weakness in sound insulation in the low-frequency range has also become apparent. In addition to experimental studies, a theoretical model capable of predicting the performance of many design variations is needed. Such a model requires mechanical and material parameters of all components. Those parameters are usually frequency dependent and may not be measured easily.

In this paper we quantify the vibration-damping ability of a sand-sawdust mixture. According to the listening tests performed on a group of people in [1], the LTFS based on

Figure 1 has better sound insulation than a solid concrete slab of 150 mm with carpet and a suspended ceiling. The sand-sawdust in the upper layer of the system improves the acoustic performance. In order to put the theoretical model to real use, the mechanical properties of the components, the sand-sawdust layer in particular, need to be quantified. The effects of the sand-sawdust layer is included in the model as a damping coefficient or an imaginary part of the stiffness of the upper layer. We use the experimental data and the theoretical model to show that the damping can be modelled as a linear function of frequency. It is commonly believed that LTFSs cannot perform as well as concrete counterparts. Blazier and DuPree [2] claimed that it was impossible to build a practical LTFS, which could satisfy an average resident. However, the prediction model used in that paper was unsuited for LTFSs.

We use the classical theories of elastic plates and beams to represent the panels, joists and battens (see [3] and [4]). Furthermore the theory of room acoustics is used to include the sound transmitted via the cavity, using the Helmholtz equation. The displacement of individual components and the acoustic pressure are expressed using the Fourier series over the two dimensional rectangular shape of the structure. Our method of using the Fourier series is an extension of the methods in [5, 6, 7, 8, 9], which deal with ribbed plates with simpler designs. We note that the

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representation of the vibration field using the Fourier series is different from more often used modal analysis with FEM or SEA. We are not directly computing the eigenfunctions or eigenvalues of the system, rather we compute the displacement of the structure (floor and ceiling surfaces) at each frequency. As a result we are able to compute the details of the mode shapes at every frequency, and thus able to compare the theoretical model and the experimental measurements precisely.

The performance of the model is evaluated by its ability to predict particularly the first few resonant frequencies and amplitudes. We have found that our model can predict the first three resonant frequencies within 1.0 Hz and their amplitudes within 3.0 dB. The resonances of the structure are not distinct after the third resonance. Although the model can predict the decaying rate of the vibration at the frequencies above 80 Hz, it cannot exactly determine the amplitude at a given frequency. We suspect that inhomogeneity and the uncertainties in the structure begin to affect the behaviour. Thus, it may not be possible to completely determine the vibration level at this frequency range. The effects of the sand-sawdust in the upper layer is included in the model as the loss factor, which is the imaginary part of the rigidity of the upper layer. The loss factor here is dependent on the frequency, so that the sand-sawdust dampens the vibration more as the frequency increases. The effects of the fibre-infill in the cavity is similarly included as the imaginary part of the mass density and the speed of sound in the Helmholtz equation.

We start with two examples of LTFSs in section 2. Brief descriptions of the designs and the construction methods are given. In section 3, we introduce the differential equations for the vibration of the individual components. Section 4 gives the procedure to determine the parameters for sand-sawdust, glassfibre fill, and rubber clips. A brief discussion on advantages of using the sand-sawdust mixture is given in section 5. We conclude the paper with some summarizing remarks in section 6.

2. Examples of LTFSs

This section shows two designs of LTFSs in Figures 2 and 3. The upper layer is modified from a single plywood to a sand-sawdust filled-layer in order to evaluate the damping effects of the sand-sawdust layers. The design of Figure 3 with a sand-sawdust layer has given the best sound insulation performances in both physical vibration measurements and subjective listening tests.

During the experimental programme in [1], altogether 26 LTFSs were built and tested. Each of those designs had small variations from the designs shown in either Figures 2 or 3. The experimental designs were determined based on the numerical simulations (see [10]) and the building practicality. For example, the simulations prior to the experiments had shown that increasing the cavity depth would have had little effects on the vibration level of the ceiling. Thus the cavity depth was not changed for the designs studied here. The theoretical simulations also showed that



Figure 1. Depiction of an LTFS. The upper layer has a sandsawdust mixture enclosed by plywood panels and timber battens. The joists are laminated timber beams. The ceiling panels are attached to the joists by furring channels and resilient rubber clips. The coordinate system later used in mathematical modelling is shown with the x and y axes.



Figure 2. Cutaway schematics of an LTFS with a single plywood upper layer. The joists span 7.0 m, and the width is 3.2 m.

the rubber clips and the furring channels on the ceiling reduced the vibration levels of the ceiling considerably, and thus they were always included in the designs. The descriptions of the commercial products used in these LTFSs are given in Appendix A1.

2.1. Single plywood upper layer

Figure 2 shows the basic design of a common joist floor, which has a plywood upper layer, supporting timber joists and a ceiling. This design has been used to determine the parameters for the resilient clips, the furring channels and the glassfibre wool in the cavity. The modelling procedure for these three components will be given in subsections 3.2 and 4.2.

2.2. Sand-sawdust upper layer

The design in Figure 3 gives the lowest average vibration level. Furthermore it has been found to be the best performing design in terms of the sound insulation perceived by a group of listeners in the room below the structure. This LTFS has 90 mm deep battens screwed to the upper plywood of the floor (at right angles to the joists) with another layer of plywood on the battens.

3. Theoretical Modelling

The model is constructed using the partial differential equations for elastic plates (Kirchhoff plates) and beams (Euler beams). These are the equations for the vertical displacement of the components (see Table I). In addition to elastic components, the model included the acoustic pressure in the cavity, which satisfies the Helmholtz equation. The equations are then coupled by the forces acting on each component that are dependent on the designs. Subsection 3.1 give the differential equations for the displacement of a thin elastic plate and Euler beams. In section 3.2 the equations are coupled using the contact conditions, spring connections and slippage displacement. Subsection 3.3 shows the method of solution using the Fourier series expansion of the displacement of each component. The modelling method for more generalized layered structures is shown in [10] in detail, which however did not specifically explain how to model the sand-sawdust layer.

3.1. Equations for the components

Our model is made up of the following components (see Figure 3):

- 1. Upper layer with two plywood boards, timber battens and sand-sawdust.
- 2. Timber joists.
- 3. Cavity filled with glassfibre wool.
- 4. Resilient rubber clips.
- 5. Furring channels attached to the ceiling.
- 6. Ceiling panel.

The origin of the coordinate system is placed at a corner of the upper layer. The joists are spanned along the x-axis. The z-axis points downward so that the upper layer is at z = 0 and the ceiling is at z = d, where d is the depth of the cavity. The top layer is excited by time-harmonic forcing with radial frequency $\omega = 2\pi f$ at (x_0, y_0) , where f is the frequency in Hz, and thus the whole system is driven at the same frequency f. The displacement of a plate for example is then given by **Re** $\left[w_{\mathbf{u}}(x, y) e^{j\omega t}\right]$ for $(x, y) \in [0, A] \times [0, B]$ where A and B are the width and length of the LTFS. Note that the velocity of the top surface is given by **Re** $[j \omega w_u(x, y) e^{j \omega t}]$. Table I gives the summary of the notations (displacement, mass density, thickness and Young's modulus) for the elastic compon ents. Note that the mass density of the upper layer includes all components, for example, the sand-sawdust, plywood panels and timber battens.

The displacement of the two plates w_u and w_c satisfies the following thin plate equation (see [4]).

$$(D_{\mathbf{u}}\nabla^{4} - m_{\mathbf{u}}\omega^{2}) w_{\mathbf{u}}(x, y) = F\delta(x_{0}, y_{0}) - P_{\mathbf{u}}(x, y)$$

$$- p(x, y, 0) \qquad (1)$$

$$(D_{\mathbf{c}}\nabla^{4} - m_{\mathbf{c}}\omega^{2}) w_{\mathbf{c}}(x, y) = P_{\mathbf{c}}(x, y)$$

$$+ p(x, y, d), \qquad (2)$$



Figure 3. Cutaway schematics of an LTFS with a sand-sawdust upper layer. The length of the structure is changed to 5.5 meters.

Table I. Notations for the elastic plates and beams.

	disp.	dens.	thick.	Mod.
upper layer	$w_{\mathbf{u}}(x, y)$	mu	h _u	E_{u}
timber battens	$w_{\mathbf{t}}(x_i, y)$	0	h_{t}	E_{t}
joists	$w_{\mathbf{j}}(x, y_{\mathbf{j}})$	mj	h_{j}	E_{j}
furring channels	$w_{f}(x_{i}, y)$	0	hf	E_{f}
ceiling	$w_{c}(x, y)$	m_{c}	h_{c}	$E_{\mathbf{c}}$

where *F* is the external force amplitude, $P_{\mathbf{u}}$ and *p* denote the force from the attached joists and the acoustic pressure from the cavity, respectively. The localized forcing is expressed by the Dirac delta function $\delta(x_0, y_0) = \delta(x-x_0, y-y_0)$. We note that ten forcing points are chosen randomly to compute the mean value of $w_{\mathbf{u}}$ in the numerical simulations. The derivation of the above equations can be found in structural acoustics text books such as [3, 11]. In equation (2), $P_{\mathbf{c}}$ and p(x, y, d) denote the force from the battens and the acoustic pressure, respectively. The ceiling is made up of two layers of plasterboard in this case. We again simplify the model by assuming that the ceiling is a single plate with two layers of varying stiffness. The differential operator is defined by

$$\nabla^4 = \frac{\partial^4}{\partial x^4} + 2\frac{\partial^4}{\partial x^2 \partial y^2} + \frac{\partial^4}{\partial y^4}.$$

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The flexural rigidity, $D_{\mathbf{u}}$, for the upper layer is computed by $E_{\mathbf{u}}h_{\mathbf{u}}^3/12(1-v^2)$, where $E_{\mathbf{u}}$, $h_{\mathbf{u}}$ and v are the Young's modulus, thickness and the Poisson ratio, respectively. The Young's modulus of the upper layer has to be adjusted because of the additional sand-sawdust layer. Thus we model the upper layer as one plate with stiffening beams (representing battens). This upper layer plate has a bending stiffness equivalent to two plates separated by a gap. Damping by the sand-sawdust is included as the imaginary part of the stiffness, denoted by $\delta_{\mathbf{s}}$. We have the adjusted stiffness

$$D_{\mathbf{u}} = \frac{E\left(h_{\mathbf{u}}^{3} - h_{\mathbf{t}}^{3}\right)}{12(1 - \nu^{2})} \left(1 + j\,\delta_{\mathbf{s}}\right),\tag{3}$$

where h_t is the thickness of the timber battens. The neutral plane of deformation is assumed to be at the middle of the upper layer. The mass of the timber battens in the upper layer is neglected because they are much lighter than the rest of the upper layer. Thus we have the following equation for the timber battens in the upper layer.

$$E_{\mathbf{t}}I_{\mathbf{t}}\frac{d^{4}}{dy^{4}}w_{\mathbf{t}}(x_{i},y) = P_{\mathbf{t}}(i,y), \quad i = 1, 2, \dots, S_{\mathbf{t}}, \quad (4)$$

where I_t is the moment of inertia of the battens. The locations of the timber battens are given by x_i , $i = 1, 2, ..., S_t$, where S_t is the number of battens. The force $P_t(i, y)$ comes from the two plywood layers. This is a simplification of the actual design which has the ribbed double-leaf plate configuration. We have made this compromise because the upper layer is thin and light compared to the rest of the structure.

We set the number of joists S_j and located at $y = y_j$, $j = 1, 2, ..., S_j$. Then the displacement of *j*th joist, $w_j(x, y_j)$ satisfies the Euler beam equation

$$\left[E_{\mathbf{j}}I_{\mathbf{j}}\frac{\mathrm{d}^{4}}{\mathrm{d}x^{4}}-m_{\mathbf{j}}\omega^{2}\right]w_{\mathbf{j}}\left(x,y_{j}\right)=P_{\mathbf{j}}(x,j)$$
(5)

for $j = 1, 2, ..., S_j$. The moment of inertia I_j is computed by $h_j^3 d_j/12$, where d_j is the width of the beams. On the right hand side, P_j denotes the total force applied on the joists from the upper layer ad the ceiling attachments.

The furring channels are light and thin, and thus we assume that they only give additional stiffness to the ceiling. The elasticity of the rubber clips determines the coupling force P_j , which is given in detail in subsection 3.2. The equations for the furring channels are

$$E_{\mathbf{f}}I_{\mathbf{f}}\frac{\mathrm{d}^{4}}{\mathrm{d}y^{4}}w_{\mathbf{f}}(x_{i},y) = P_{\mathbf{f}}(i,y) \tag{6}$$

for $i = 1, 2, ..., S_f$. The locations of the furring channels (and the rubber clips) are given by x_i , $i = 1, 2, ..., S_f$, where S_f is the number of furring channels. Note that the same notation x_i is used for the timber battens in the upper layer, however the indices are distinguished by the notations S_f and S_f .

The cavity air acts as primary path of sound at low-frequencies, and thus it must be included in the model. The acoustic pressure, p(x, y, z), in the cavity satisfies the Helmholtz equation (see [12]).

$$\left(\nabla^2 + \mu^2\right) p(x, y, z) = 0,$$
 (7)

where $\mu = \omega/c$, and *c* is the speed of sound of the cavity air. The cavity walls are assumed to be acoustically hard, that is the normal derivatives on the walls are zero. Thus we have the following boundary conditions for the normal derivatives of *p* on the side-walls of the cavity.

$$p_x(0, y, z) = 0, \ p_x(A, y, z) = 0, p_y(x, 0, z) = 0, \ p_y(x, B, z) = 0.$$
(8)



Figure 4. Depiction of the modelling regime of the junctions between components.

We have not considered the effects of sub-diving the cavity by the joists, because the wavelengths at low-frequency vibration are long compared to the spacing of the joists (600 mm centres). The coupling conditions between the plates (upper and ceiling) and the air are given by

$$p_z(x, y, z) = \omega^2 \rho w_{\mathbf{u}}, \quad p_z(x, y, d) = \omega^2 \rho w_{\mathbf{c}}, \quad (9)$$

where ρ is the mass density of the air. The mass density and μ in the cavity are modified to account for the sound damping by the glassfibre fill.

3.2. Conditions for the junctions and the cavity

The equations derived in the previous subsection need to be coupled using the physical conditions at the junctions. The coupling conditions will be expressed using the conditions for the displacement and the force. For example, the displacement of the upper layer and the joists at their junctions will be the same, and the force acting on the ceiling will be the same reaction of the force on the furring channels.

There are various methods of modelling the junctions. However, it is common to model the junction either completely loose (see [5, 6, 7, 13]) or continuous elastic material (see [14, 15]). We here add some resistance between the upper plate and the joist to model the condition that is neither loose nor rigid. A detailed study of slippage is given in [10].

The force P_u in equation (1) includes the direct vertical force and the resistance force due to the slippage depicted in Figure 4. The force acting on the upper plate due to the slippage between the upper layer and *j*th joist is

$$P_{\text{slip}(\mathbf{u})}(x,j) = \frac{\sigma h_{\mathbf{u}} \left(h_{\mathbf{u}} + h_{\mathbf{j}} \right)}{4} \frac{\mathrm{d}^2}{\mathrm{d}x^2} w_{\mathbf{u}}(x,y_j), \quad (10)$$

where σ is the slippage rigidity constant. Then we have

$$P_{\mathbf{u}}(x, y) = -\sum_{j=1}^{S_{j}} \left\{ P(x, y_{j}) + P_{\text{slip}(\mathbf{u})}(x, j) \right\}$$
$$\cdot \delta \left(y - y_{j} \right), \qquad (11)$$

where $P(x, y_j)$ is the direct vertical force acting on the upper layer.

The rubber connectors give

$$P_{ij} = \tau \left(w_{\mathbf{u}}(x_i, y_j) - w_{\mathbf{c}}(x_i, y_j) \right)$$
(12)

for $i = 1, 2, ..., S_f$, $j = 1, 2, ..., S_j$ where τ is the spring constant, which is assumed to be same for all *i* and *j*. In section 4, a frequency dependent τ is tested, though we have settled with a constant value in the end. Thus the force terms on the right hand side of equation (6) is given by

$$P_{\mathbf{f}}(i, y) = -P_{\mathbf{c}}(x_i, y) + \sum_{j=1}^{S_j} P_{ij}\delta(y - y_j).$$

The above force is coupled to the ceiling on the right hand side of equation (2). The force acting on jth joist is

$$P_{j}(x, j) = P(x, y_{j}) - P_{slip(j)}(x, j) - \sum_{i=1}^{S_{1}} P_{ij}\delta(x - x_{i})$$
(13)

where the force on the joist $P_{\text{slip}(j)}$ due to the slippage is given by

$$P_{\text{slip}(\mathbf{j})}(x, j) = \frac{\sigma h_{\mathbf{j}} \left(h_{\mathbf{u}} + h_{\mathbf{j}}\right)}{4} \frac{d^2}{dx^2} w_{\mathbf{u}} \left(x, y_j\right).$$

The force acting on the ceiling in equation (2) comes from the furring channels. Hence we have

$$P_{\mathbf{c}}(x, y) = \sum_{i=1}^{S_{\mathbf{f}}} P_{\mathbf{f}}(i, y) \,\delta(x - x_i).$$

3.3. Fourier series expansion

We use the Fourier expansion method to compute the solution of the system of equations given in the previous section. The orthogonal sine-basis functions are given by

$$\begin{cases} \phi_m(x) = \sqrt{2/A} \sin k_m x, \\ \psi_n(y) = \sqrt{2/B} \sin \kappa_n y \end{cases}$$
(14)

for m, n = 1, 2, ..., where the wavenumbers are $k_m = \pi m/A$ and $\kappa_n = \pi n/B$. These modes satisfy the simply-supported conditions, that is, there is zero displacement and zero bending moment at the boundary of the structure.

The displacement of the upper plate and the ceiling can be expressed by

$$w_{\mathbf{u}}(x, y) = \sum_{m,n=1}^{N} c_{mn}^{\mathbf{u}} \phi_m(x) \psi_n(y), \qquad (15)$$

$$w_{c}(x, y) = \sum_{m,n=1}^{N} c_{mn}^{c} \phi_{m}(x) \psi_{n}(y), \qquad (16)$$

where $\{c_{mn}^{\mathbf{u}}\}\$ and $\{c_{mn}^{\mathbf{c}}\}\$ are complex valued coefficients and N is the truncated number of modes we use to compute the solutions. Substituting equations (15) and (16) to the equations of the individual components and using the orthogonality will lead to a system of algebraic equations for $\{c_{mn}^{\mathbf{u}}, c_{mn}^{\mathbf{c}}\}$, in other words we will have an equation for the vectors, $c_{\ln ameu}$ and $c_{\mathbf{c}}$ of $\{c_{mn}^{\mathbf{u}}\}$ and $\{c_{mn}^{\mathbf{c}}\}$, respectively. We have assumed that the upper layer and the joists are always in contact. Then we have $w_{\mathbf{u}}(x, y_j) = w_{\mathbf{j}}(x, j)$. Similarly the furring channels and the ceiling are assumed be always in contact, and thus we have $w_{\mathbf{c}}(x_i, y) =$ $w_{\mathbf{f}}(i, y)$. We then need to find the coefficients $\{c_{mn}^{\mathbf{u}}, c_{mn}^{\mathbf{c}}\}$ to find the displacement everywhere in the structure. Note that the series solutions are truncated to N terms for computation. It was found that N = 20 to be sufficient for the frequency range studied here.

The acoustic pressure in the cavity space is also expressed using the Fourier expansion, which uses cosine basis function,

$$\alpha_m(x) = \sqrt{\frac{2}{A}} \cos k_m x, \quad \beta_n(x) = \sqrt{\frac{2}{B}} \cos \kappa_n y$$

for m, n = 0, 1, ... These modes satisfy the acousticallyhard wall conditions at the cavity walls. The acoustics pressure is

$$p(x, y, z) = \sum_{m,n=0}^{N} \left\{ \Gamma_{mn}^{(1)} e^{\gamma_{mn} z} + \Gamma_{mn}^{(2)} e^{-\gamma_{mn} z} \right\} \cdot \alpha_m(x) \beta_n(y),$$
(17)

where $\{\Gamma_{mn}^{(1)}, \Gamma_{mn}^{(2)}\}\$ are the coefficients to be determined and the wavenumbers are defined by $\gamma_{mn} = \sqrt{k_m^2 + \kappa_n^2 - k^2}$. Finally, substituting the expansions given by equations (15), (16) and (17) into the partial differential equations derived in the previous section will give us a system of linear equations for $\{c_{mn}^{u}, c_{mn}^{c}, \Gamma_{mn}^{(1)}, \Gamma_{mn}^{(2)}\}\$. The matrices corresponding to the above formulations are given in Appendix A2.

The root-mean-square velocity of the ceiling surface, which will be compared against the experimental results, is computed by the integration of the square of the velocity given by equation 18. The integral was computed numerically once the values of $v(x, y) = j \omega w_c(x, y)$ over the ceiling surface had been computed.

$$\sqrt{\langle |v|^2 \rangle} = \sqrt{\int_0^B \int_0^A |v(x, y)|^2 \, \mathrm{d}x \, \mathrm{d}y}.$$
 (18)

4. Determining the parameters for the cavity, ceiling attachments and the sandsawdust layer

This section shows how the parameters for the cavity and the sand-sawdust mixture are determined. For the cavity we need the flow resistivity of the glassfibre filling. For the sand-sawdust, we need the damping coefficient. Once a parameter is determined, no further adjustments are made when another component is added to the model and the real structure. This is to avoid arbitrary parameter-fitting whenever the design is changed. We start with a brief description of the experimental setup for measuring the ceiling vibration. More details can be found in [1, 10].

4.1. Experimental setup

On each floor an electrodynamic shaker was used to provide a vertical force on the upper floor surface. The shaker was connected to the floor through a wire stinger and a reference force transducer. The stinger is there to ensure that only vertical forces are transmitted in the floor, while the force transducer lets us know how much force is sent into the floor. The shaker body was mounted on a beam which straddled the floor and rested on supports which sat on the concrete collar surrounding the floor. Vibration isolation of the beam from the concrete collar was provided by very resilient pads made of polyester fibre infill. The shaker was driven with pseudo-random signal with a bandwidth from 10 Hz to 500 Hz, for a duration of 2 seconds (to achieve a frequency resolution of 0.5 Hz). The position on the floor was selected so that the low-frequency modes would be excited. Only one position on each floor was chosen. It is often useful to select two or more positions on a structure to ensure a sufficient number of modes are excited, and to act as a check for results. However, in this case, it would have taken too long to do two complete vibration response scans of each floor.

A scanning laser vibrometer (Polytec PSV 300) was used to measure the velocity of the floor upper surface vibration and the ceiling vibration in a direction which is normal to the surfaces. A grid with a spatial resolution of 10-14 cm was used to obtain a map of the surface velocity of the floor and ceiling relative to the input force; both amplitude and phase information was recorded at each frequency. The laser vibrometer measurement equipment was connected to the force transducer so that the recorded surface vibration is normalised with respect to the force applied. Therefore the results shown in the following sections are the transfer function between the velocity and the input force. This enables direct comparison to the theoretical modelling which also uses a unit amplitude input force. The signal sent to the shaker was matched with the sampling time of the laser vibrometer software. This ensured minimal spectral leakage and a frequency resolution of 0.5 Hz. The measurement results shown in the graphs in this paper were generated by taking a root mean square average of the measured surface velocity across the whole ceiling of the floor systems.

4.2. Cavity fibre infill

Here we explain briefly how the flow resistivity of the glassfibre (see Figure 5) is determined using the simulations of the system in Figure 2. This design was chosen because the system only includes the joists and the panel components, whose theoretical model has already been determined in [10]. Thus the effects of the cavity should show clearly in the modelling.

The effect of the fibrous infill in the cavity is modelled using a complex propagation constant μ in equation (7). This complex propagation constant consists of a real phase constant and an imaginary attenuation constant, and can be determined from existing models of sound propagation in



Figure 5. Photograph of the ceiling attachment system showing the steel furring channel and the resilient rubber clips. The fibreglass infill between joists is also shown here.

porous media. An overview and application of such models is given in a number of recent text books, in particular [16] and [17]. Most porous media models assume that the porous media forms a rigid frame, within which the air moves. This rigid-frame assumption is valid for high frequencies. In our case, however, we consider low frequencies, where the wavelength may be larger than the porous media thickness, and thus we cannot necessarily assume that the porous media is not in motion. The Biot theory describing wave motion in an elastic porous media can be used in this more complex case, where the frame is not assumed to be rigid (see [16]). However, for the purposes of this paper and our model, we did not go to the complexity of using the Biot theory, but based our cavity model on the empirical model and results presented in Appendix C of [17].

When the wavelength of sound in the fibrous infill is less than the thickness of the infill we can assume that the fibrous infill is rigid and unmoving, and μ in the cavity is modified to

$$\mu = \frac{2\pi f}{c_0} \sqrt{\frac{1 - (1 - \gamma)\eta}{1 + \sigma}},$$
(19)

where c_0 is the speed of sound in air, γ is the ratio of specific heats for the gas (=1.4 for air), and

$$\eta = 0.592 a (X_1) + j b (X_1),
\sigma = a(X) + j b(X),$$
(20)

where $X = \rho_0 f/R_1$ and $X_1 = 0.856\rho_0 f/R_1$. R_1 is the flow resistivity, and ρ_0 is the air density. The formulas for *a* and *b* are given in Appendix A3.

When the wavelength of sound in the fibrous infill is greater than the thickness of the infill, the fibrous infill can no longer be regarded as rigid and unmoving. As a consequence the absorption is reduced compared to the short wavelength case. Appendix C of [17] gives results of expected transmission loss for porous layers in the lowfrequency range when the porous material can no longer be regarded as rigid. These results suggest that the transmission loss (in dB) tends to zero as the frequency tends



Figure 6. The root-mean-square velocity of the ceiling of the system shown in Figure 2. (a) the flow resistivity is constant 7200 Rayls m^{-1} , (b) the flow resistivity is modelled by equation (22). The theoretical modelling results are shown by the dashed line, and the measurements are shown by the solid line.

to zero. Thus there is a smooth transition to the highfrequency range for transmission loss against the log of the frequency. We also note that the flow resistivity of a porous medium is related to the pressure drop across the porous medium, which does not only apply to static flow but also to oscillations at low frequencies (see [11]). Therefore, at low frequencies the transmission loss across the porous material ΔL_p is approximately related to the flow resistivity R_1 by

$$10^{\Delta L_p/20} \propto R_1. \tag{21}$$

As a result of these considerations, we may define a modified flow resistivity \widetilde{R}_1 for low-frequencies, which is related to the otherwise determined transmission loss across a thickness of porous material. \widetilde{R}_1 is then inserted into equation (19) to produce μ . We use the results in Appendix C of [17] for the transmission loss in porous materials at low frequencies and assume that \widetilde{R}_1 is zero at f = 0 and linearly increasing until the wavelength of sound in the cavity λ_m is equal to the thickness of the fibrous infill in the cavity d at which point $\widetilde{R}_1 = R_1$,

$$\widetilde{R}_{1} = \begin{cases} R_{1}f/f_{K} & \text{for } f < f_{K}, \\ R_{1} & \text{for } f \ge f_{K}, \end{cases}$$
(22)

where f_K is the frequency at which $\lambda_m = d$ and is numerically determined using equation (19). For the flow resistivity of 7200 Rayls/m and thickness of 300 mm, we have $f_K = 980$ Hz.

The floor shown in Figure 2 has a ceiling consisting of two layers of 13 mm plasterboard screwed to furring channels at 600 mm centres, which are attached to every other

joist (i.e. at 800 mm centres) through resilient clips (see appendix A1 for material properties). Our model assumes that there is a ceiling clip on every joist, which is compensated for by reducing the stiffness of each clip by a suitable amount. The cavity is 358 mm deep and is filled with two layers of 150 mm sound control type fibreglass. The experimental test version of this floor had ceiling furring channels 35 mm deep and a calculated stiffness of 11000 Nm². The ceiling furring resilient clips have a measured stiffness of 220000 Nm^{-1} with a 130 N constant load, and a loss factor of about 0.1. The cavity infill has a flow resistivity of 7227 Raylm⁻¹ and a density of 12 kgm⁻³. Figures 6a,b show the difference between the constant and linear modelling of the flow resistivity. Setting the flow resistivity constant gives poor agreement at low-frequencies with the experiment as shown in Figure 6a. Figure 6b shows that the linear model gives better agreement at the lowfrequencies, though the behaviour at higher frequencies changes little. The theoretical modelling results are obtained by simulating ten random excitation locations on the top layer.

4.3. Sand-sawdust layer

Measurements were made on a floor with a sand and sawdust layer in the upper floor (Figure 3). 90 mm deep battens separate two layers of 15 mm plywood, and in-filled between the battens is an 85 mm layer of paving sand combined with sawdust with an 80/20 mix ratio by volume. Mixing sawdust in with the sand gives better damping than using sand alone, and helps prevent the sand compacting over time.

We compare experimental measurements on this floor with the model results to see the effect of having a large amount of mass, stiffness and damping on the upper surface of the floor. The model is not able to represent the upper surface exactly as built, but is able to model an equivalent single upper surface plate with rib stiffeners running perpendicular to the joists. Experimental results (namely the position of the second resonance peak) tell us about the stiffness of this plate. We estimated the loss factor of the upper surface of the floor for frequencies above 100 Hz by measuring the amplitudes of the surface vibrations of the floor and the rate of decay as the vibrations propagated from the shaker excitation point. We averaged the decay measurements performed in directions parallel and perpendicular to the joist directions to produce a single value of the loss factor estimate. For frequencies 100 Hz and above we therefore determined the loss factor to be between 0.4 and 0.8. This seems to be a large value for loss factor, but such values are reported by Richards and Lenzi [18] for pure sand on plates, and by Yanagida et al. [19] for binary powder mixtures (including sand and rubber powder). At the fundamental bending frequency of the system we determined the loss factor to be 0.1 by measuring the width of the fundamental peak.

Sun *et al.* [20] and Lin *et al.* [21] showed that the damping due to the sand layer on a plate is frequency dependent. Thus we expect the same for the mixture and try several



Figure 7. Comparison of the theoretical (dashed) and measured (solid) root-mean-square velocity of the ceiling when δ is constant as in equation (23).



Figure 8. Comparison of the theoretical (dashed) and measured (solid) root-mean-square velocity of the ceiling when δ is given by equation (24).



Figure 9. Comparison of the theoretical (dashed) and measured (solid) root-mean-square velocity of the ceiling when δ is given by equation (25).

functions of frequency to model the damping.

$$\delta_{\rm s} = 0.4, \tag{23}$$

$$\delta_{\rm s} = \frac{0.8}{200} f, \tag{24}$$

$$\delta_{\rm s} = \frac{0.4}{200} f. \tag{25}$$



Figure 10. Comparison of the experimental measurements of the root-mean-square velocity of the ceilings of the LTFSs with sand-sawdust (85 mm thick) and a floating gypsum concrete upper layer with equivalent mass and significantly less bending stiffness.

These functions are simplifications of a not very well understood phenomenon. Figures 7, 8 and 9 show the comparison between the models with the constant damping and the frequency dependent damping given by equations (23), (24) and (25), respectively. The modelling results in Figures 8 and 9 show that the frequency dependent damping for the sand-sawdust layer gives better agreement with the measurements, around the first two resonant peaks (between 10 and 30 Hz). There are, however, significant deviations from the measurement results for some of the resonances between 30 and 50 Hz. This is probably due to inaccuracies of the estimated loss factor of the sand and sawdust fill.

5. Discussion: Advantages of using sandsawdust

We emphasize the effectiveness of the sand-sawdust layer in damping vibration in Figure 10 by comparing the performance of a system with a layer of floating gypsum concrete (see Figure 12) as the upper layer. The gypsum concrete upper layer system has comparable mass (81 kgm^{-2}) to the sand-sawdust upper layer system (mass 113 kgm⁻²), but rather less bending stiffness (3.4×10^4 Nm cf. 1.2×10^5 Nm). Figure 11 shows the comparison between the gypsum concrete and sand-sawdust systems when the upper layer of the sand-sawdust system is thinner (45 mm gap between plywood layers, and 40 mm of sand-sawdust mix in the gap). The weight of the upper layer of this thinner sand-sawdust system is 56 kgm⁻². The comparison in Figures 10 and 11 shows that the damping contribution by the sand-sawdust cannot be replicated by simply adding equivalent mass and stiffness. The sandsawdust layer dampens the vibration above 60 Hz more effectively than the concrete upper layer when the weight of the upper layer is similar. One should also note that we obtained good results with the sand-sawdust upper floor system when compared to the gypsum concrete even though the gypsum concrete is a floating raft (on a resilient layer),



Figure 11. Comparison of the experimental measurements of the root-mean-square velocity of the ceilings of the LTFSs with sand-sawdust (40 mm thick) and a floating gypsum concrete upper layer with equivalent bending stiffness and significantly greater mass.



Figure 12. Cutaway schematics of the LTFS with a floating gypsum concrete upper layer. This has timber I-beams for the joists, which are lighter than LVL.

and the sand-sawdust floor is not – the top plywood layer is screw-fixed to the bottom plywood layer via battens.

Although this paper is about low-frequency vibrations, standard ISO tapping-machine impact sound insulation tests were conducted during the experimental programme in [1]. We found that the concrete and sand-sawdust systems had the same $L_{n,w}$ value of 52 dB when they had similar upper layer stiffness, whereas the thicker (similar weight) sand-sawdust system had an $L_{n,w}$ value of 48 dB.

Figures 13 and 14 show numerical simulations of various stiffness and mass densities of the upper layer. The mass density and the stiffness are varied in order to confirm that the damping by the sand-sawdust cannot be achieved by replacing it with simple mass. This fact is consistent with the comparison of the experimental results shown in Figure 10. Both simulations in Figures 13 and 14 show that the increase in mass and stiffness certainly lowers the vibration level above 80 Hz. Furthermore, it takes an impractical amount of mass and stiffness to achieve performance comparable to that of sand-sawdust. For example, the gypsum concrete upper layer (81 kgm⁻²) used



Figure 13. Numerical simulations of the root-mean-square velocity of the ceiling with various upper layer bending stiffness.



Figure 14. Numerical simulations of the root-mean-square velocity of the ceiling with various upper layer mass densities.

in Figure 10 would be nowhere heavy or stiff enough to achieve the dotted curves in Figures 13 and 14. When observing the vibrations of the floor upper surface, flexural standing waves could be seen in the gypsum concrete topping, whereas the flexural waves in the sand-sawdust floor upper did not propagate far enough to form standing waves [1]. Measurements by others on gypsum board show the loss factor to be approximately 0.02 (see [22]).

6. Summary

This paper has demonstrated the potential of the vibration damping ability of the sand-sawdust layer in LTFSs. The sand-sawdust layer achieved greater performance than an equivalently massive concrete layer. Thus the benefit of using the sand-sawdust in LTFSs has been confirmed. In order to quantify the damping, a mathematical model is used to determine the loss factor of the sand-sawdust layer. It has been found that using a loss factor as a linear function of frequency gives us a good agreement between theoretical predictions and experimental measurements. In addition, a modelling regime for the fibre-infill in the cavity space has been studied. The flow-resistivity of the infill as a linear function of frequency has given us a good agreement between the theory and the experiments. Although the theories of fibrous media and granular media tell us that there is a complex interaction between the damping and the frequency, simple linear functions have been chosen so that the mathematical model is kept simple and practical.

Appendix

A1. Material parameters

- 1. Panel products
 - 15 mm 5-ply Ecoply F11 plywood: Manufacturer's nominal Density = 560 kgm⁻³, nominal static bending stiffness 2360 Nm² along face grain, 684 Nm² perpendicular to face grain assuming 10.5 GPa along-grain wood stiffness. Dynamic measurements from one sample showed that along-grain wood stiffness was 13 GPa. Apparent measured dynamic bending stiffness along face grain (from floor measurements) is equivalent to homogeneous material with Young's modulus from 12 to 14 GPa. Vibration loss factor of material assumed to be 0.03.
 - 13 mm GIB Noiseline (gypsum) plasterboard: Manufacturer's nominal density = 962 kgm⁻³. Dynamic Young's modulus = 3.7 GPa. Measured vibration loss factor = 0.013. Supplied by Winstone Wallboards Ltd.
 - Gypsum top layer: USG Levelrock 3500 PS, presanded gypsum concrete. Manufacturers nominal density = 1920 kgm^{-3} . Nominal Young's modulus = 6.6 GPa.
- 2. Joists
 - Carter-Holt-Harvey (CHH) Hyspan LVL (laminated veneer lumber): Manufacturer's nominal density = 620 kgm⁻³, nominal static Young's modulus = 13.2 GPa. Apparent dynamic Young's modulus from measurements = 14.5 GPa to 15.5 GPa. Assumed vibration loss factor = 0.03.
 - 300 mm CHH Hybeam I-beam (HJ300-63): Manufacturer's nominal linear density = 4.4 kgm^{-1} , nominal static bending stiffness = 1111000 Nm^2 . Assumed vibration loss factor = 0.03.
- 3. Infill materials
 - 150 mm Tasman Insulation Mid-floor Silencer: Measured sample flow resistivity = 7227 Rayls m⁻¹. Density = 12 kgm⁻³.
 - Sand/sawdust mix (80/20): Density = 1210 kgm^{-3} .
- 4. Ceiling fixtures
 - RSIC clip: Dynamic Stiffness at 20 Hz under 130 N load (approx equiv to 25 kgm⁻² ceiling surface density) = 220000 Nm⁻¹. Loss factor = 0.1.
 - Gib Rondo Batten: Estimated (from measurements) bending stiffness when attached to plasterboard = 11000 Nm².

A2. Formulation of the matrices

This section shows the formulation of the linear system of equations for the coefficients of the Fourier expansion of the displacement of the floor, the ceiling and the acoustic pressure. Although most of the derivation here has been presented in [10], the reader may find these formulas useful when implementing our model.

The equations for the coefficients are formulated as matrix and vector equations using the vector representation of the coefficients, that is, the coefficients are denoted by the following column vectors

$$c_{\mathbf{u}} = \begin{pmatrix} c_{11}^{\mathbf{u}} \\ c_{12}^{\mathbf{u}} \\ \vdots \\ c_{NN}^{\mathbf{u}} \end{pmatrix}, c_{\mathbf{c}} = \begin{pmatrix} c_{11}^{\mathbf{c}} \\ c_{12}^{\mathbf{c}} \\ \vdots \\ c_{NN}^{\mathbf{c}} \end{pmatrix}$$

and

$$\boldsymbol{\Gamma}_{1} = \begin{pmatrix} \Gamma_{00}^{(1)} \\ \Gamma_{01}^{(1)} \\ \vdots \\ \Gamma_{NN}^{(1)} \end{pmatrix}, \boldsymbol{\Gamma}_{2} = \begin{pmatrix} \Gamma_{00}^{(2)} \\ \Gamma_{01}^{(2)} \\ \vdots \\ \Gamma_{NN}^{(2)} \end{pmatrix}$$

The assembled equation of the vector $(c_u, c_c, \Gamma_1, \Gamma_2)$ is

$$\begin{bmatrix} M_1 & M_{12} & M_{13} & M_{14} \\ M_{12} & M_2 & M_{23} & M_{24} \\ M_{31} & 0 & M_3 & M_{34} \\ 0 & M_{42} & M_{43} & M_4 \end{bmatrix} \begin{pmatrix} c_{\mathbf{u}} \\ c_{\mathbf{c}} \\ \Gamma_1 \\ \Gamma_2 \end{pmatrix} = \begin{pmatrix} f \\ 0 \\ 0 \\ 0 \end{pmatrix}.$$

For example, the elements of M_1 are given by equation (A1).

$$M_{1} = \begin{bmatrix} \ddots & 0 & 0 \\ 0 & D_{0} \left(k_{m}^{2} + \kappa_{n}^{2}\right)^{2} - m_{0}\omega^{2} & 0 \\ 0 & 0 & \ddots \end{bmatrix} \\ + \begin{bmatrix} \tau \sum_{i,j=1}^{S_{i},S_{1}} \phi_{1} \left(x_{i}\right)\psi_{1} \left(y_{j}\right)\phi_{1} \left(x_{i}\right)\psi_{1} \left(y_{j}\right) \\ & \vdots \\ & \ddots & \ddots & \vdots \\ \cdots & \tau \sum_{i,j=1}^{S_{i},S_{1}} \phi_{N} \left(x_{i}\right)\psi_{N} \left(y_{j}\right)\phi_{N} \left(x_{i}\right)\psi_{N} \left(y_{j}\right) \end{bmatrix} \\ + \begin{bmatrix} \left(E_{1}I_{1}k_{m}^{4} - m_{1}\omega^{2}\right)\sum_{j=1}^{S_{1}}\psi_{1} \left(y_{j}\right)\psi_{1} \left(y_{j}\right) \\ & \vdots \\ & 0 \\ & \ddots & & \vdots \\ \cdots & \left(E_{1}I_{1}k_{m}^{4} - m_{1}\omega^{2}\right)\sum_{j=1}^{S_{1}}\psi_{N} \left(y_{j}\right)\psi_{N} \left(y_{j}\right) \end{bmatrix} \\ + \begin{bmatrix} Hk_{m}^{2}\sum_{j=1}^{S_{1}}\psi_{1} \left(y_{j}\right)\psi_{1} \left(y_{j}\right) \\ & \vdots \\ & 0 \\ & \ddots & & 0 \\ & \ddots & & \vdots \\ & \cdots & 0 \\ & \ddots & & \vdots \\ & \cdots & Hk_{m}^{2}\sum_{j=1}^{S_{1}}\psi_{N} \left(y_{j}\right)\psi_{N} \left(y_{j}\right) \end{bmatrix}$$
 (A1)

$$H = \frac{\sigma \left(h_{\mathsf{u}} - h_{\mathsf{j}} \right) \left(h_{\mathsf{c}} + h_{\mathsf{j}} \right)}{2}.$$

The sub-matrix M_{12} represents the spring connection by the rubber clips, and is same as the second matrix in the above equation with an opposite sign.

The other off-diagonal sub-matrices M_{13} , M_{14} , M_{23} , and M_{24} represent the interaction between the cavity air and the top/bottom surfaces given by equations (1) and (2). For example, M_{13} is due to the pressure at z = 0 and given by equation (A2).

$$\begin{bmatrix} \int_{0}^{A} \int_{0}^{B} \alpha_{1}\beta_{1}\phi_{1}\psi_{1}dxdy & \int_{0}^{A} \int_{0}^{B} \alpha_{1}\beta_{2}\phi_{1}\psi_{1}dxdy \\ \int_{0}^{A} \int_{0}^{B} \alpha_{1}\beta_{1}\phi_{1}\psi_{2}dxdy & \int_{0}^{A} \int_{0}^{B} \alpha_{1}\beta_{2}\phi_{1}\psi_{2}dxdy \\ \vdots & \cdots \\ \int_{0}^{A} \int_{0}^{B} \alpha_{1}\beta_{1}\phi_{N}\psi_{N}dxdy & \cdots \\ \cdots & \int_{0}^{A} \int_{0}^{B} \alpha_{1}\beta_{N}\phi_{1}\psi_{1}dxdy \\ \vdots & \vdots \\$$

 $\cdots \int_0^A \int_0^B \alpha_N \beta_N \phi_N \psi_N dx dy \bigg]$

Note that $M_{13} = M_{14}$. Similarly, the matrices M_{23} and M_{24} can be derived from equation (9) at z = d. The matrices M_{31} and M_{42} are diagonal matrices of $\omega^2 \rho$ from equation (9), and the matrices M_3 , M_{34} , M_{43} , and M_4 are due to the left hand side of equation (9).

The vector f due to the forcing is given by

$$f = \begin{pmatrix} F\phi_1(x_0)\psi_1(y_0) \\ F\phi_2(x_0)\psi_1(y_0) \\ \vdots \\ F\phi_N(x_0)\psi_N(y_0) \end{pmatrix}.$$

There may be more efficient way of assembling the matrices. We have not considered it, because the largest matrix is $4N^2$ by $4N^2$, N = 20, which can be computed using an average personal computer.

A3. Flow resistivity

The functions a(X) and b(X) in equation (20) are defined by

$$\begin{split} a(X) &= \frac{T_3 \left(T_1 - T_3\right) T_2^2 - T_4^2 T_1^2}{T_3^2 T_2^2 + T_4^2 T_1^2},\\ b(X) &= \frac{T_1^2 T_2 T_4}{T_3^2 T_2^2 + T_4^2 T_1^2}, \end{split}$$

where

$$\begin{split} T_1 &= 1 + 9.66X, \ T_2 = X + 0.0966X^2, \\ T_3 &= 2.537 + 9.66X, \ T_4 = 0.1591 \left(1 + 0.7024X \right). \end{split}$$

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